

National Aeronautics and Space Administration



# Technical Feasibility Panel for the Human Space Flight Study

National Research Council

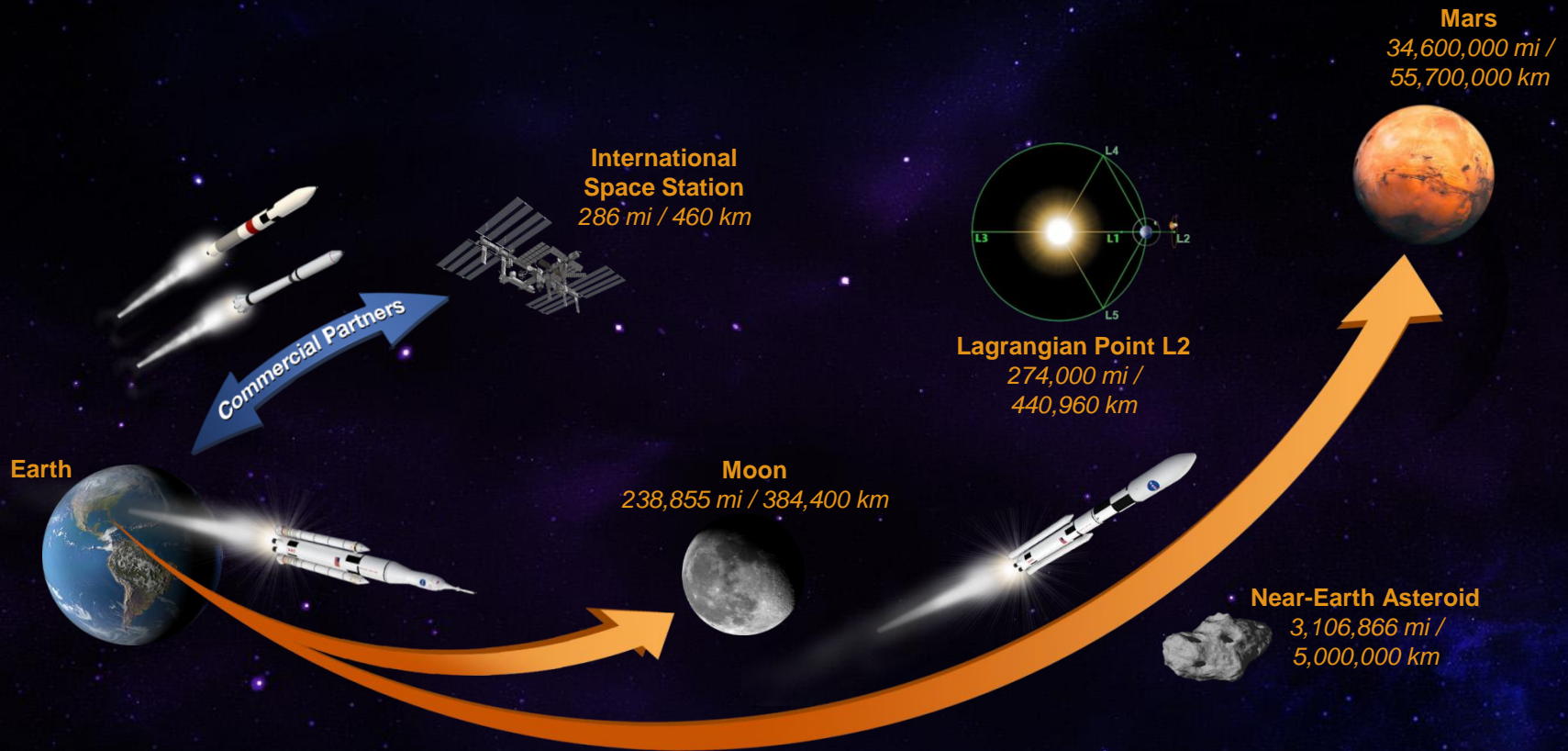
**Dan Dumbacher**  
**February 4, 2013**



## Overview

- Human Space Exploration
- Exploration Systems Development Overview
- Space Launch Systems
- Orion Multi-Purpose Crew Vehicle
- Ground Systems Development and Operations
- Missions
- Challenges

# The Future of Exploration



## Human Spaceflight Capabilities



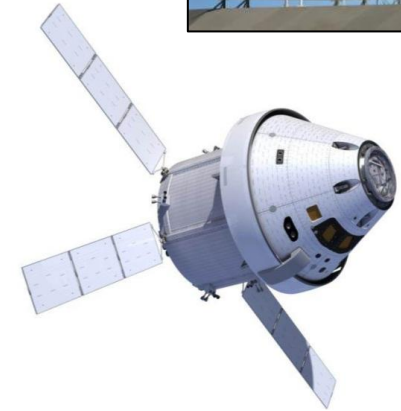


# Exploration Systems Development



**These programs will develop the launch and spaceflight vehicles that will provide the initial capability for crewed exploration missions beyond LEO.**

- **Space Launch System (SLS) Program:**
  - Initial capability: 70 tonnes (t), 2017–2021
  - Evolved capability: 105 t and 130 t, post-2021
- **Orion Program:**
  - Initial test flight (no crew) on Delta IV in 2014 – vehicle assembly underway
  - First Orion/SLS (no crew) flight in 2017
  - First crewed Orion/SLS flight in 2021
- **Ground Systems Development and Operations (GSDO) Program:**
  - Developing launch site infrastructure to prepare, assemble, test, launch and recover the SLS and Orion flight systems



# Hardware / Test / Mission Milestones



## 2008



## 2009

Ares 1-X  
Launch

Water  
Recovery  
Training



## 2010

Attitude Control  
Motor Test

Pad abort 1  
stacking

Pad abort 1 test



## 2011

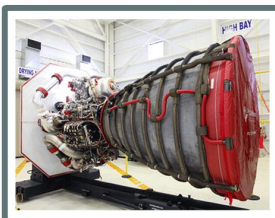
First Orion  
Environmental  
Testing

Parachute  
Tests

Solid Rocket  
Booster  
development  
motor test



## 2012



First MSA Ring Forging,  
Cudahy, Wisconsin

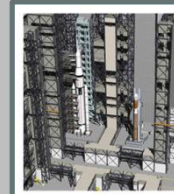
MSA Pathfinder Hardware



Exploration  
Flight  
Test-1



Final  
Assembly  
in VAB



Launch Pad B



Mobile Launch  
Platform and Tower

Exploration  
Mission-1



# LOOKING AHEAD 2013 - 2017

# The Space Launch System (SLS)



- **Affordable & Sustainable**
  - Maximum use of common elements and existing assets, infrastructure, and workforce
  - Competitive opportunities for affordability insertion
- **Safe : Human Rated**
- **Initial capability: 70 tonnes (t), 2017–2021**
  - Serves as primary transportation for Orion and exploration missions
  - Provides back-up capability for crew/cargo to ISS
- **Evolved capability: 105 t and 130 t, post-2021**
  - Can enable scientific payloads with requirements beyond commercial lift capabilities
  - Modular and flexible, sized to mission requirements
- **Liquid hydrogen and liquid oxygen propulsion system**
  - RS-25 from the Space Shuttle Program for core stage
  - Upper Stage trades in work
- **Solid rocket boosters for the initial flights**
  - Competition for follow-on boosters based on performance requirements and affordability considerations



# SLS Evolution and Block Upgrade Approach

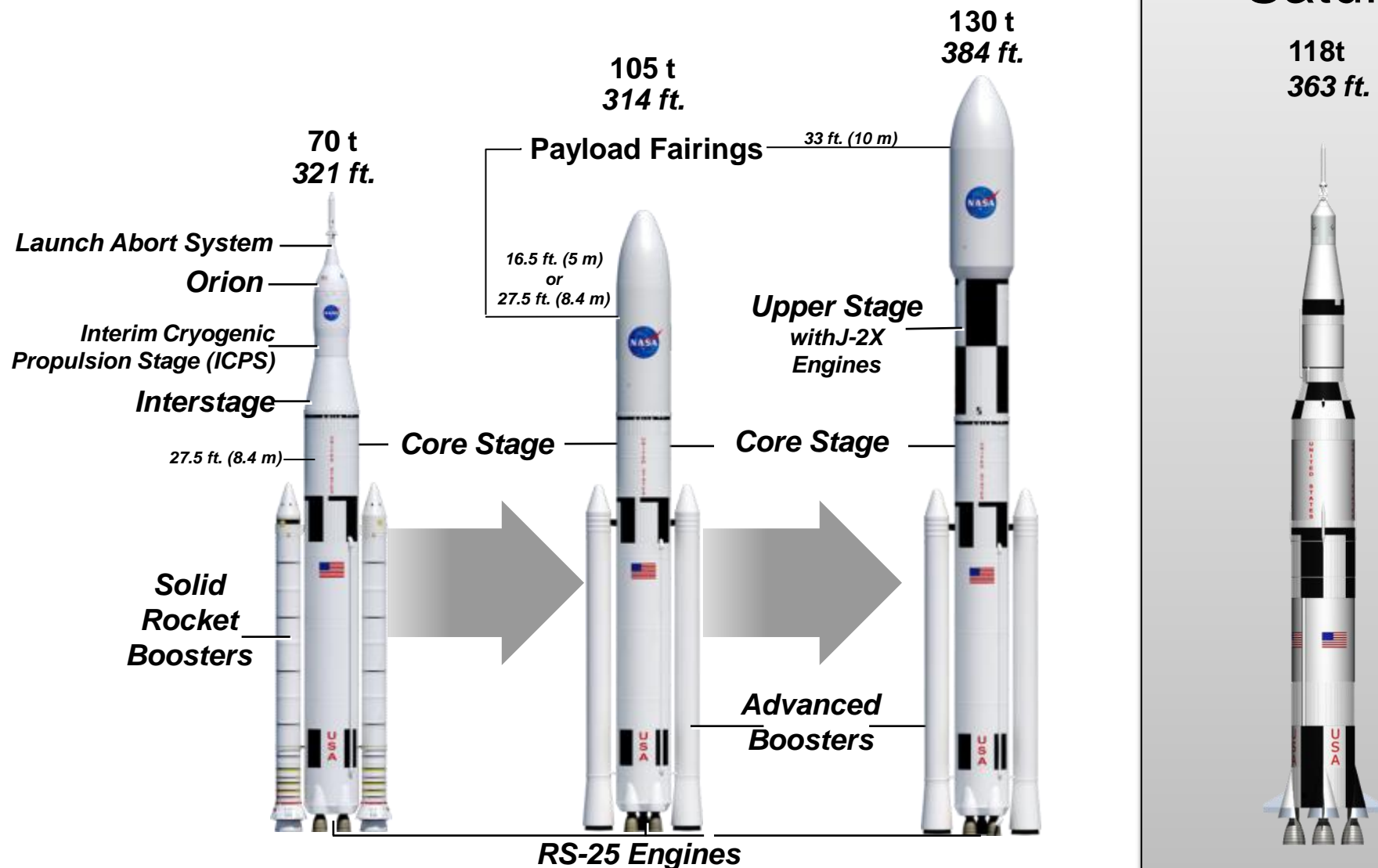


## Block 1

## Block 1A

## Block 2

## Saturn



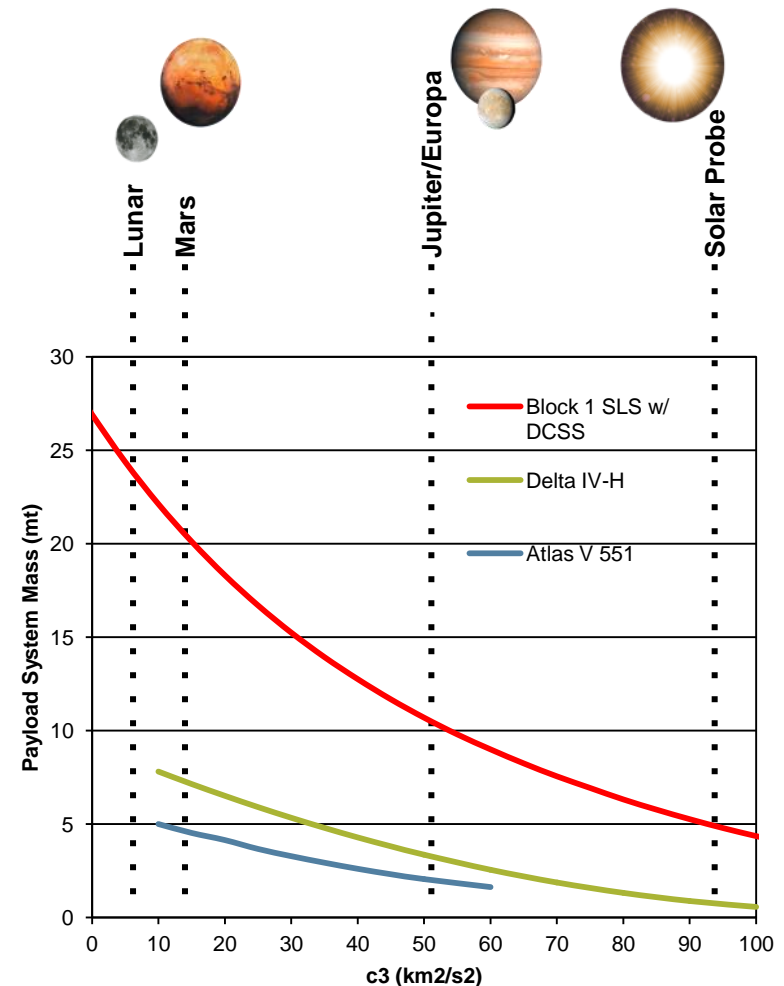


# SLS Performance Supports Deep-Space Operations



- SLS Enables Exploration Missions
  - Greater volume and mass capability/margin
    - Increased design simplicity
    - Fewer origami-type payload designs needed to fit in the fairing
  - Single launch of multiple elements means fewer launches, deployments, and critical operations
    - Simplifies on-orbit operations
    - Reduced risk
  - High-energy orbit and shorter trip times
    - Less expensive mission operations
    - Reduced risk - Maximize mission reliability via Increased lift capacity and payload margin
- SLS investment can be leveraged for other missions requiring large volume or up mass
  - Deep Space Exploration
  - Planetary Landers
  - Human Habitats
  - Great Observatories
  - Space Solar Power
  - Outer Planet Missions
  - Department of Defense/  
NRO Payloads

## SLS Block 1 C3 Performance





# SLS Accomplishments



Systems Engineering and Integration SLS model undergoes wind tunnel testing at Langley Research Center (Nov 2012)



J-2X power pack assembly hot fire test at Stennis Space Center (Nov 2012)



Multi-Purpose Crew Vehicle Stage Adapter (MSA) Pathfinder Hardware at Marshall Space Flight Center (June 2012)



Kennedy Space Center Complex 39B ready for a 2017 SLS launch (artist's concept)

## System Requirements Review/System Definition Review Completed



RS-25 Engines at Stennis Space Center Oct 2012, shown with future RS-25 Test Stand A1



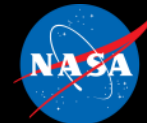
F-1 engine gas generator hot fire test at Marshall Space Flight Center, Jan 2013 – technology development for an optional Advanced Booster concept



Qualification Motor 1 casting at ATK (Oct 2012)<sub>9</sub>



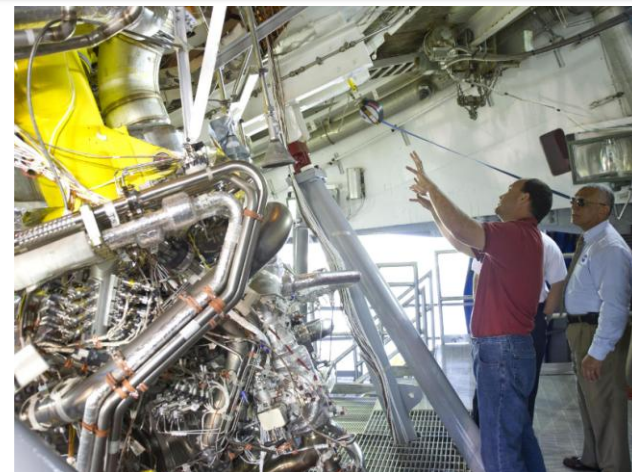
# SLS Hardware



**Stages Industry Day at Michoud Assembly Facility  
Nov 2011**



**SLS Nozzle Nose Rig**



**J-2X Upper Stage Engine at Stennis Space  
Center (SSC), April 2012**



**Solid Rocket Booster development motor test  
in Promontory, Utah, Sep 2011**



**RS-25 Core Stage Engines Stored at SSC  
Jan 2012**



**Subscale Solid Rocket Motor firing at  
Marshall Space Flight Center, March  
2012**



**MPCV Stage Adapter Production  
Major Tool and Machine, IN**

# Orion MPCV Spacecraft Overview



The Orion MPCV design divides critical functions among multiple modules to maximize the performance of the integrated spacecraft design

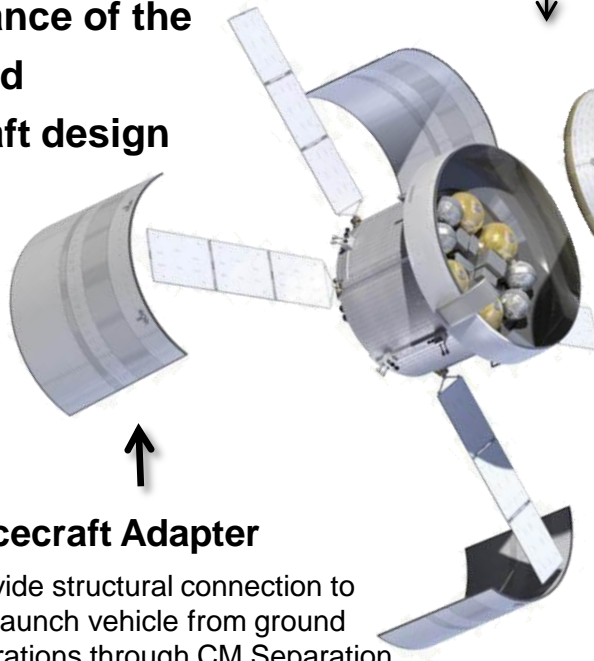
## Crew Module (CM)

- Provide safe habitat from launch through landing and recovery
- Conduct reentry and landing as a stand alone module



## Launch Abort System

- Provide protection for the CM from atmospheric loads and heating during first stage flight
- Safely jettison after successful pad operations and first stage flight



## Spacecraft Adapter

- Provide structural connection to the launch vehicle from ground operations through CM Separation
- Provide protection for SM components from atmospheric loads and heating during first stage flight

## Service Module (SM)

- Provide support to the CM from launch through CM separation to missions with minimal impact to the CM





# Orion Hardware Accomplishments



Lori Garver visits Lockheed Martin Denver to review the heatshield carrier structure



Heatshield Skeleton Assembly at Lockheed Martin in Denver



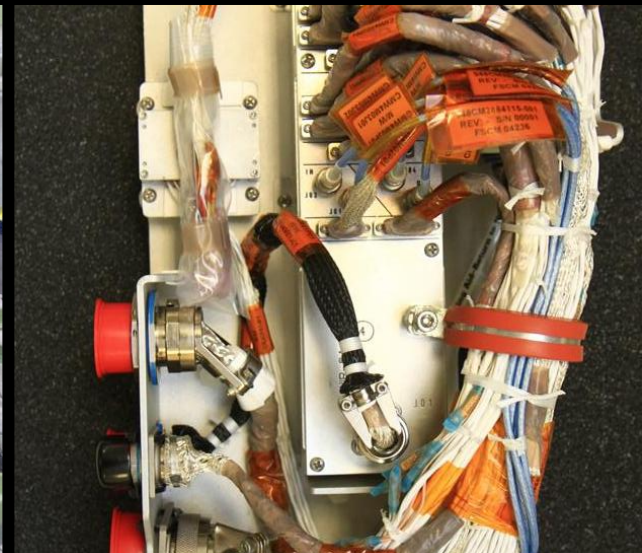
Spacecraft Adapter Jettison fairings at the Michoud Assembly Facility



EFT1 Crew Module being moved into the Birdcage tool at the Operations and Checkout Building



EFT1 Service Module panels being assembled at the Operations and Checkout Building



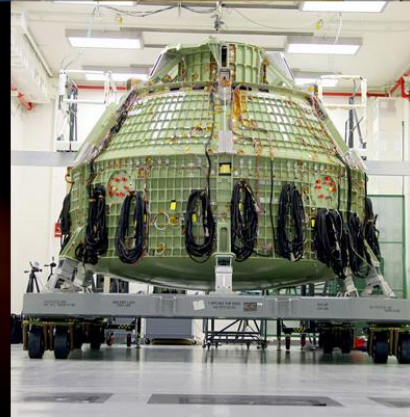
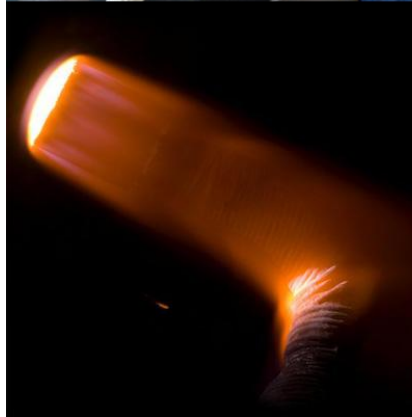
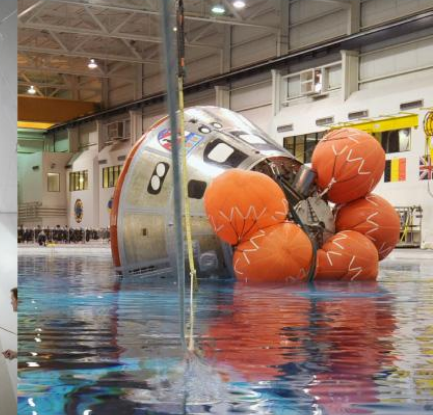
Master Data Acquisition Unit harnesses on prototype pallet fixture



# Orion Demonstration Tests Completed



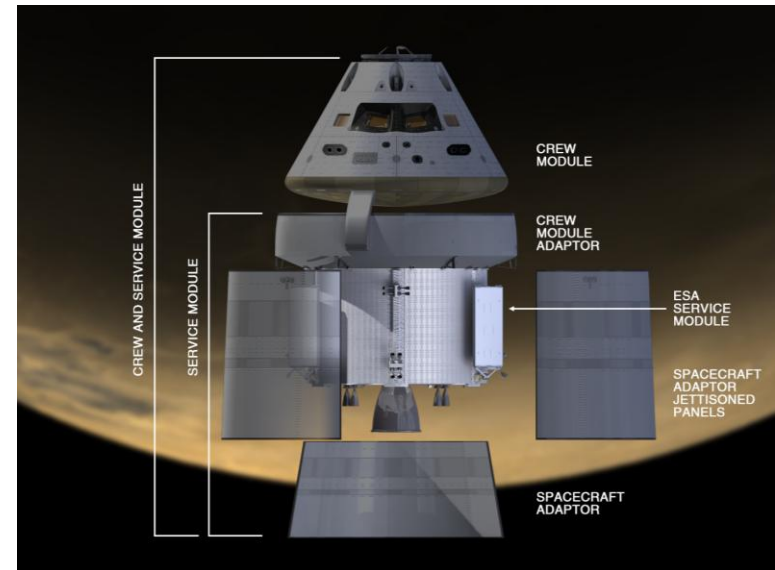
- Launch Abort System
- Parachute Drop
- Water Drop
- Human Factors - Suit
- Acoustic Vibration
- Up-righting System
- Thermal Protection System
- Landing & Recovery
- Controls Evaluation



# Orion Service Module/ESA Partnership



- NASA signed an agreement in December 2012 for the European Space Agency (ESA) to provide a service module for the Orion spacecraft's Exploration Mission-1 in 2017.
- The agreement primarily maps out a plan for ESA to fulfill its share of operational costs and additional supporting services for the International Space Station by providing the Orion service module and necessary elements of its design for NASA's Exploration Mission-1 in 2017.
- The service module will house Orion's power, thermal and propulsion systems. It will contain the in-space propulsion capability for orbital transfer, attitude control and high-altitude ascent aborts. It also will generate and store power and provide thermal control, water and air for the astronauts. It will remain connected to the crew module until just before the capsule returns to Earth.





# Ground Systems Development and Operations

Mobile Launcher Arriving at Pad 39B



# GSDO Accomplishments



## *Kennedy Space Center Evolution, 1950-2020*



## *Vehicle Integration and Launch*



## *Command Control Communications & Range*



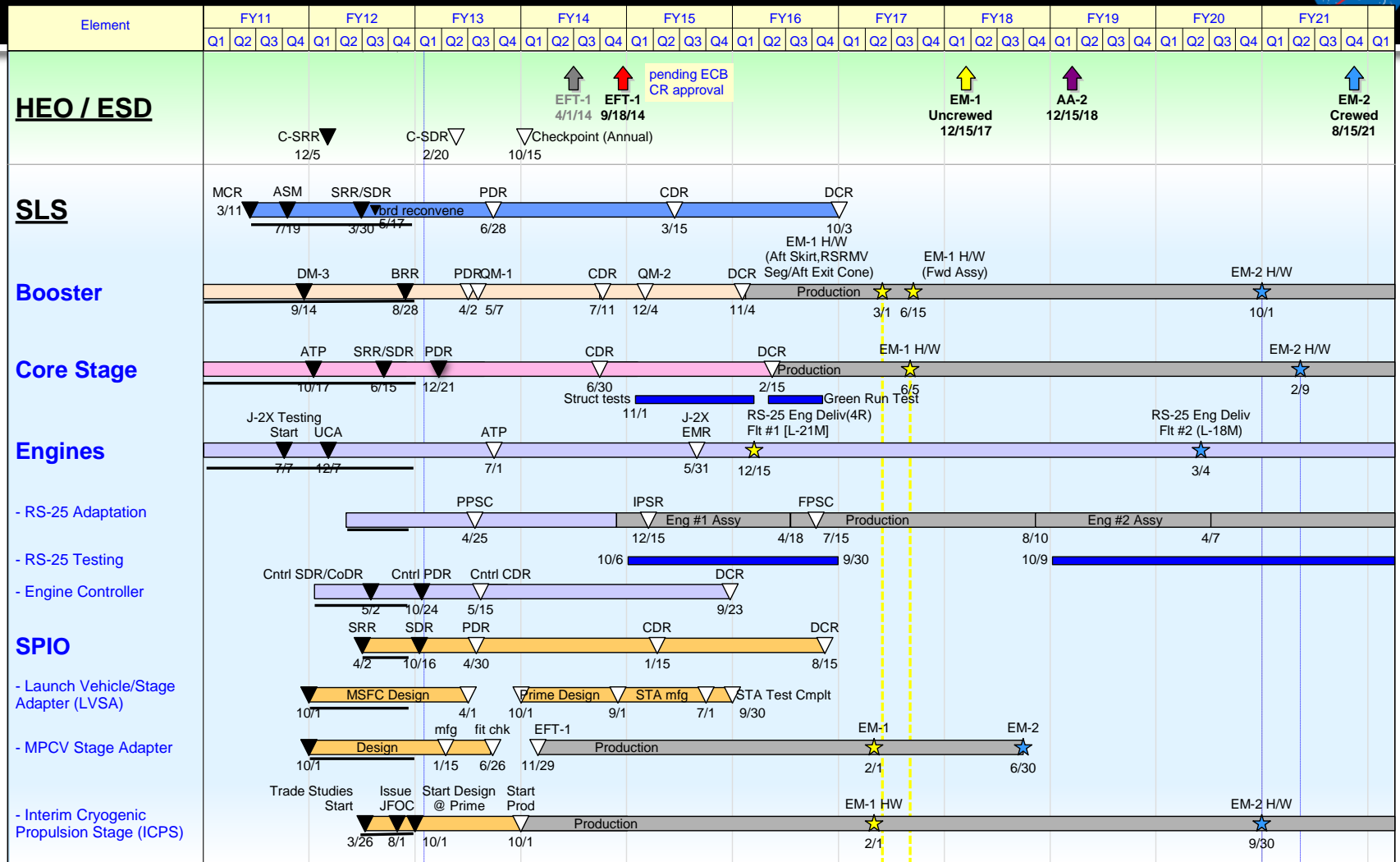
## *Offline Processing & Infrastructure*





# ESD EM-1 / EM-2 PPBE14 Integrated Schedule

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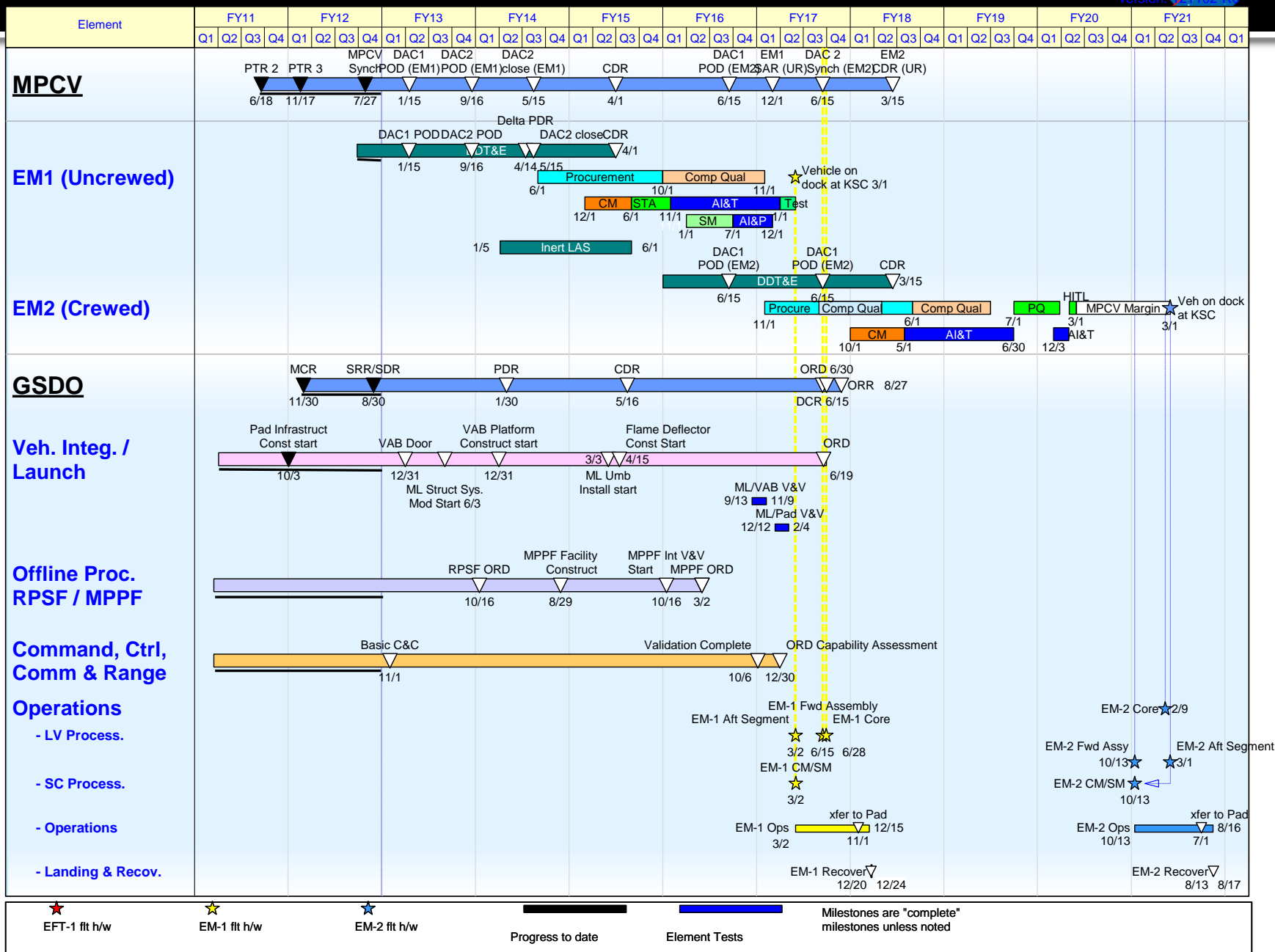


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# ESD EM-1 / EM-2 PPBE14 Integrated Schedule

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# ESD Mission Overview



## Exploration Mission One (EM-1)

First Uncrewed BEO Flight  
2017

- **Mission objectives**

- Demonstrate integrated spacecraft systems performance prior to crewed flight
- Demonstrate high speed entry (~11 km/s) and TPS prior to crewed flight

- **Mission description**

- Un-crewed circumlunar flight – free return trajectory
- Mission duration ~7 days

- **Spacecraft configuration**

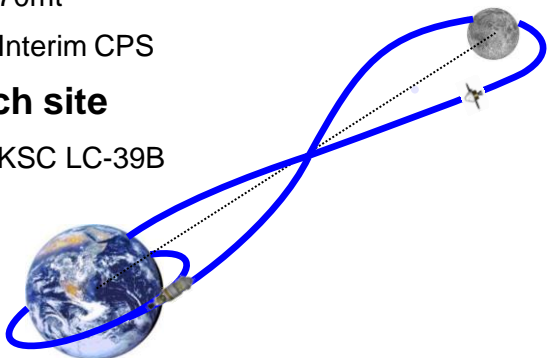
- Orion Uncrewed

- **Launch vehicle configuration**

- SLS Block 1, 5-segment RSRMV, 4 RS-25, 70mt
- Interim CPS

- **Launch site**

- KSC LC-39B



## Exploration Mission Two (EM-2)

First Crewed BEO Flight  
2021

- **Mission objectives**

- Demonstrate crewed flight beyond LEO

- **Mission description**

- Crewed lunar orbit-capable, or other destinations
- Mission duration 10-14 days

- **Spacecraft configuration**

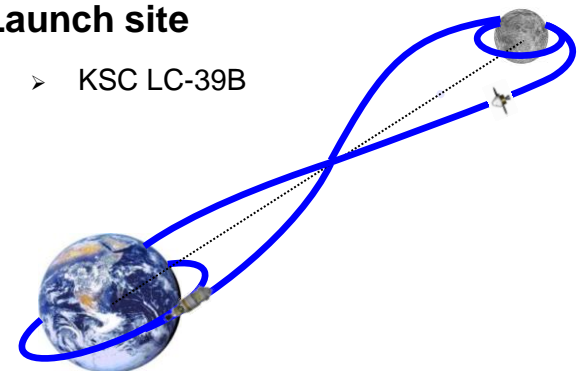
- Orion Crewed

- **Launch vehicle configuration**

- SLS Block, 5-segment RSRMV, 4 RS-25, 70mt
- Interim CPS

- **Launch site**

- KSC LC-39B



# Exploration Flight Test – 1



## EXPLORATION FLIGHT TEST ONE

OVERVIEW

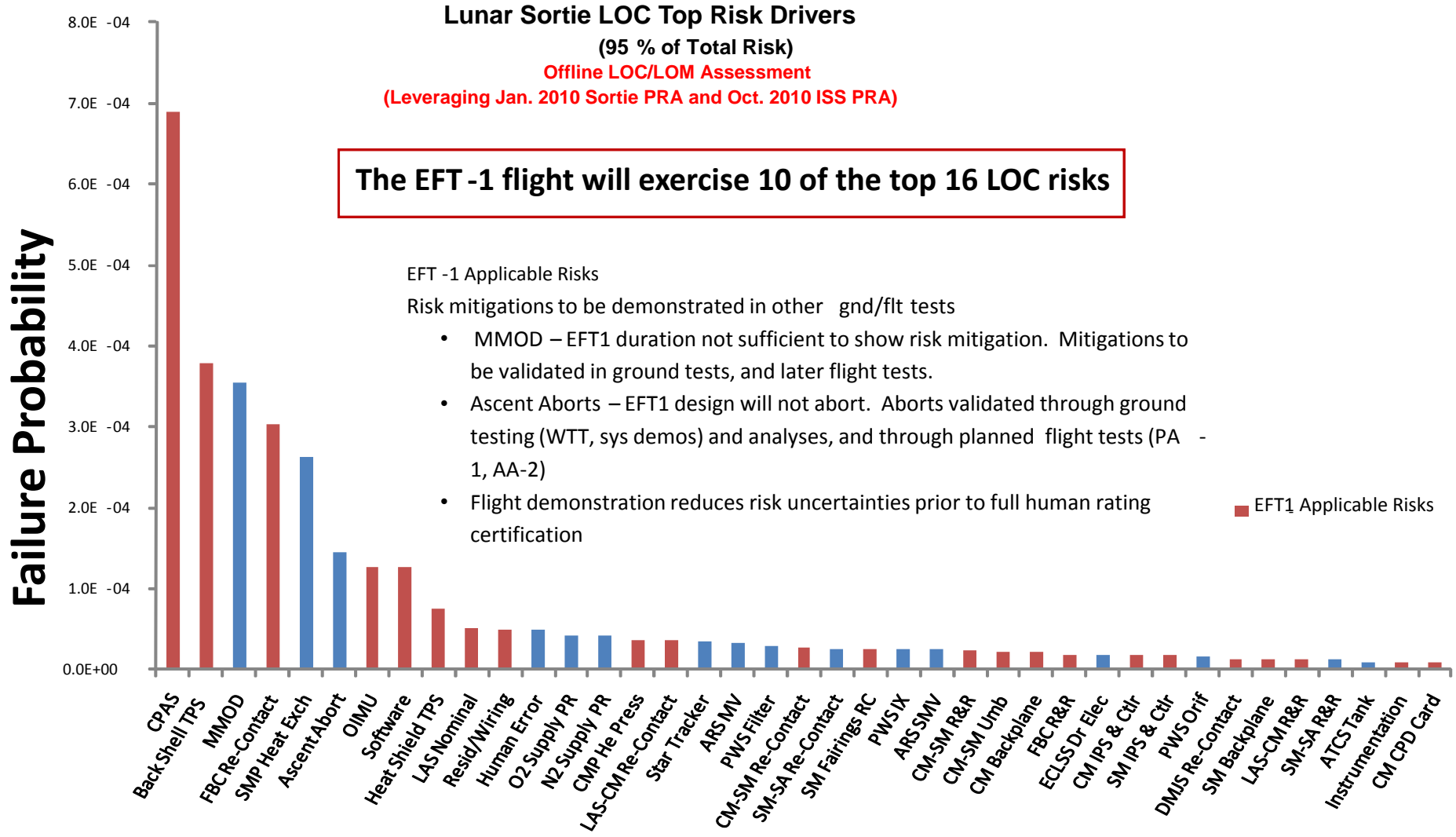
TWO ORBITS • 20,000 MPH ENTRY • 3,671 MILE APOGEE • 28.6 DEGREE INCLINATION



2014



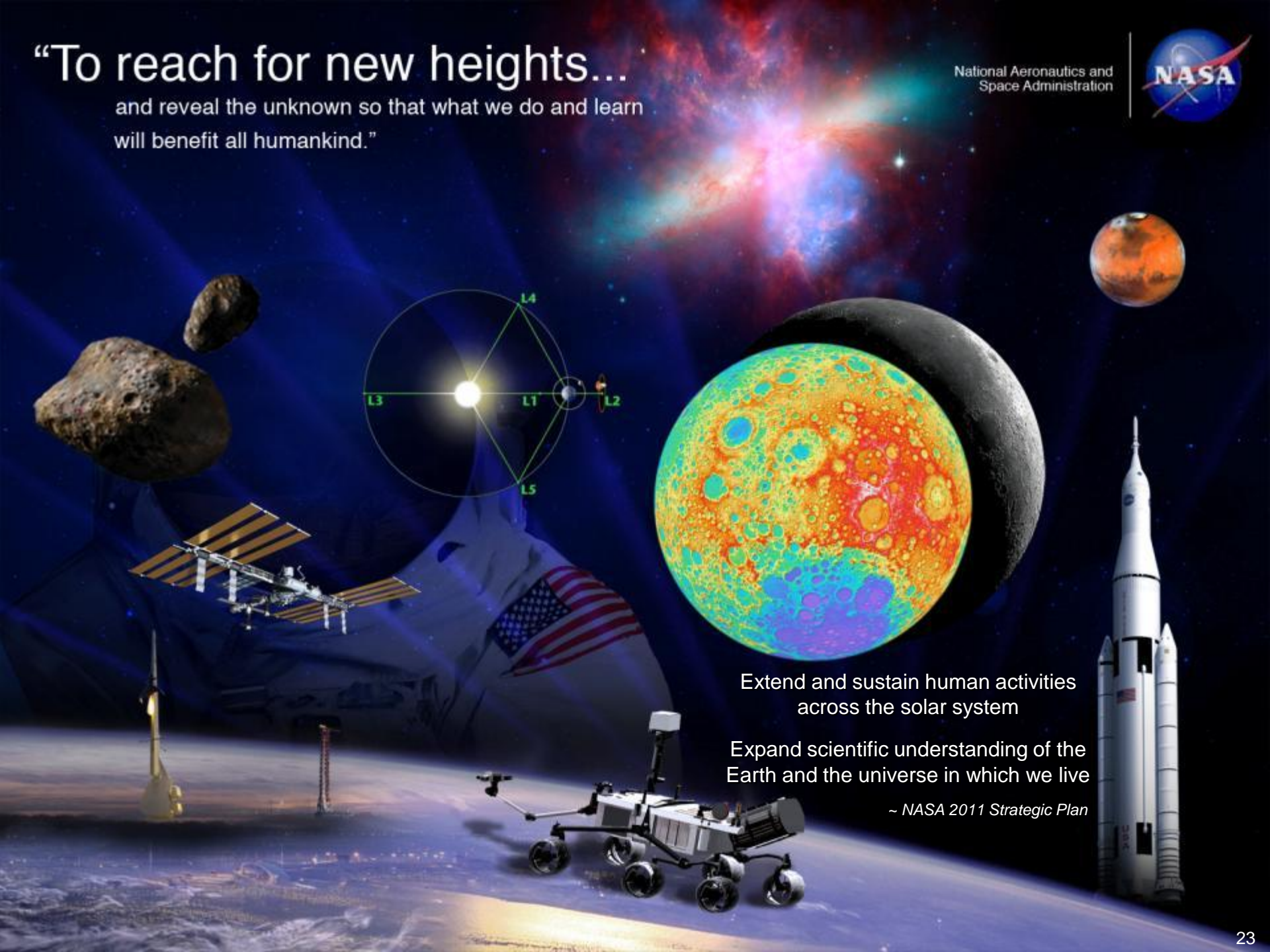
# Orion Risk Mitigation: EFT-1 exercises 10 of top 16 Loss of Crew Risks

- Significant progress has been made, and is being made, in design and development, testing, and programmatic.
- We are talking with stakeholders, independent review teams and technical authorities to ensure technical excellence in a cost and schedule constrained environment
  - NAC
  - ASAP
  - ESD Standing Review Board
  - Office of the Chief Engineer
  - Safety and Mission Assurance Office
  - Crew Health and Medical Office

“To reach for new heights...  
and reveal the unknown so that what we do and learn  
will benefit all humankind.”

National Aeronautics and  
Space Administration



Extend and sustain human activities  
across the solar system

Expand scientific understanding of the  
Earth and the universe in which we live

~ NASA 2011 Strategic Plan