

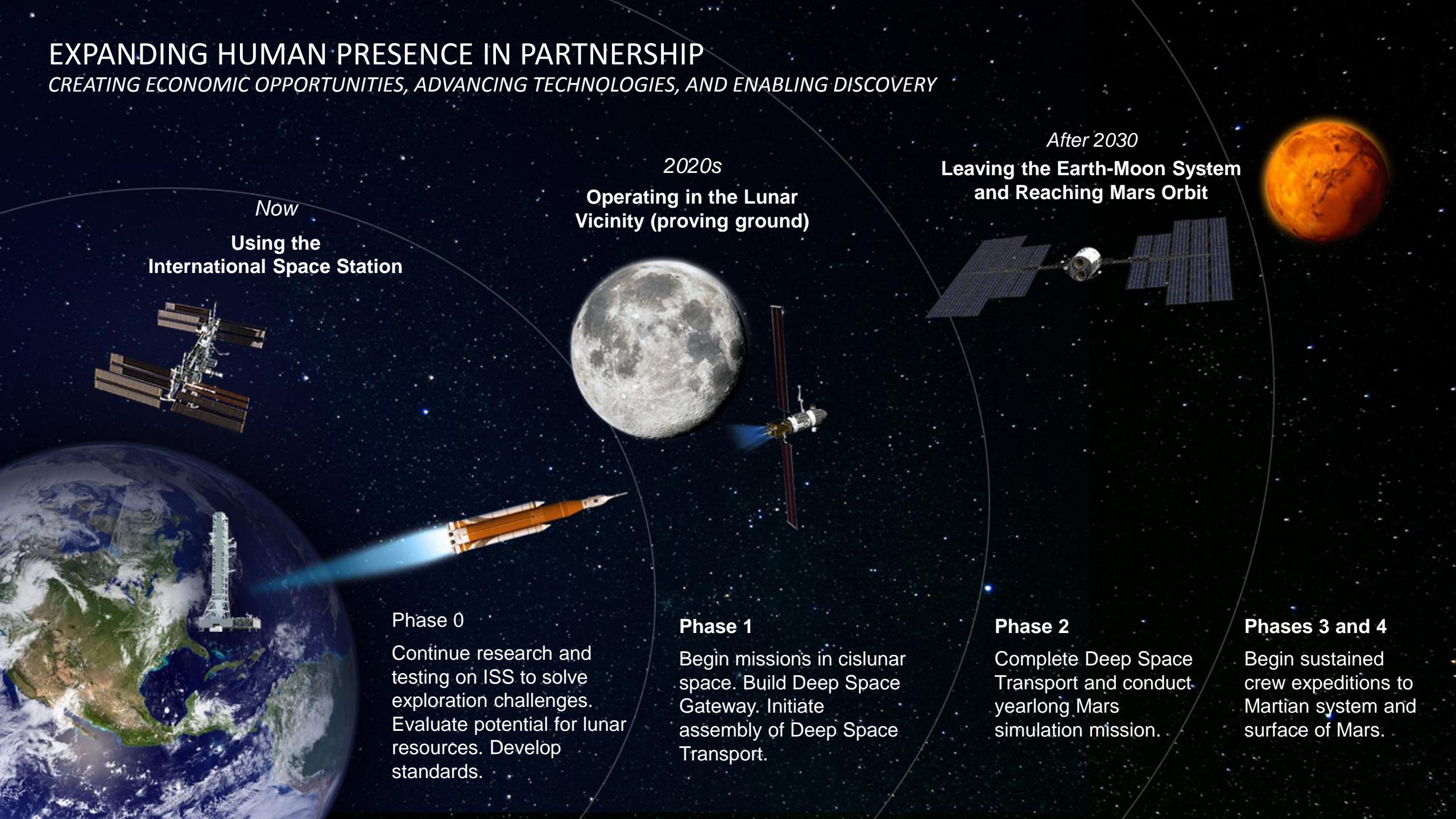


Deep Space Gateway and Journey To Mars Update

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EXPANDING HUMAN PRESENCE IN PARTNERSHIP

CREATING ECONOMIC OPPORTUNITIES, ADVANCING TECHNOLOGIES, AND ENABLING DISCOVERY



- **FISCAL REALISM**

Implementable in the near-term with the buying power of current budgets and in the longer term with budgets commensurate with economic growth;

- **SCIENTIFIC EXPLORATION**

Exploration enables science and science enables exploration; leveraging scientific expertise for human exploration of the solar system.

- **TECHNOLOGY PULL AND PUSH**

Application of high Technology Readiness Level (TRL) technologies for near term missions, while focusing sustained investments on technologies and capabilities to address the challenges of future missions;

- **GRADUAL BUILD UP OF CAPABILITY**

Near-term mission opportunities with a defined cadence of compelling and integrated human and robotic missions, providing for an incremental buildup of capabilities for more complex missions over time;

- **ECONOMIC OPPORTUNITY**

Opportunities for U.S. commercial business to further enhance their experience and business base;

- **ARCHITECTURE OPENNESS AND RESILIENCE**

Resilient architecture featuring multi-use, evolvable space infrastructure, minimizing unique developments, with each mission leaving something behind to support subsequent missions;

- **GLOBAL COLLABORATION AND LEADERSHIP**

Substantial new international and commercial partnerships, leveraging current International Space Station partnerships and building new cooperative ventures for exploration; and

- **CONTINUITY OF HUMAN SPACEFLIGHT**

Uninterrupted expansion of human presence into the solar system by establishing a regular cadence of crewed missions to cis-lunar space during ISS lifetime.



PHASE 1

Deep Space
Gateway (DSG) Concept



Phase 1 Plan

Establishing deep-space leadership and preparing for Deep Space Transport development



		Deep Space Gateway Buildup					
EM-1	Europa Clipper	EM-2	EM-3	EM-4	EM-5		
2019 - 2025						2026	
SLS Block 1 Crew: 0	SLS Block 1B Cargo	SLS Block 1B Crew: 4 CMP Capability: 8-9T	SLS Block 1B Crew: 4 CMP Capability: 10mT	SLS Block 1B Crew: 4 CMP Capability: 10mT	SLS Block 1B Crew: 4 CPL Capability: 10mT		
Distant Retrograde Orbit (DRO) 26-40 days 	Jupiter Direct 	Multi-TLI Lunar Free Return 8-21 days 	Near Rectilinear Halo Orbit (NRHO) 16-26 days 	NRHO, w/ ability to translate to/from other cislunar orbits 26-42 days 	NRHO, w/ ability to translate to/from other cislunar orbits 26-42 days 		
Gateway (blue) Configuration (Orion in grey)							

These essential Gateway elements can support multiple U.S. and international partner objectives in Phase 1 and beyond

Known Parameters:

- Gateway architecture supports Phase 2 and beyond activities
- International and U.S. commercial development of elements and systems
- Gateway will translate uncrewed between cislunar orbits
- Ability to support science objectives in cislunar space

Open Opportunities:

- Order of logistics flights and logistics providers
- Use of logistics modules for available volume
- Ability to support lunar surface missions



- **Assumptions**

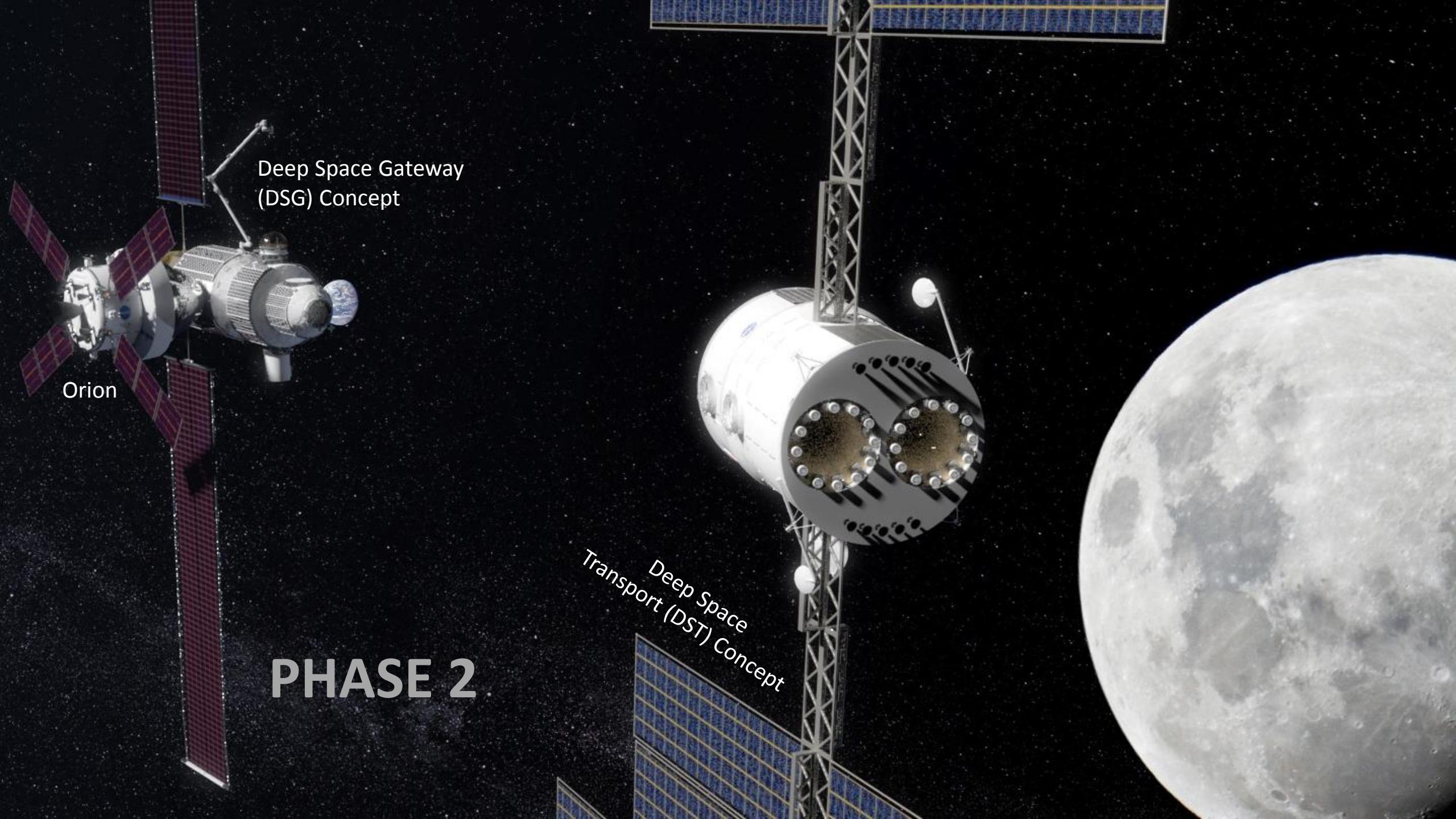
- Deep Space Gateway provides ability to support multiple NASA, U.S. commercial, and international partner objectives in Phase 1 and beyond
- The Gateway is designed for deep space environments
 - Supports (with Orion docked) crew of 4 for a minimum of 30 days
 - Supports buildup of the Deep Space Transport

- **Emphasis on defining early Phase 1 elements**

- Gateway Power Propulsion Element
- Gateway Habitat
- Logistics Strategy

- **Future work to refine later elements; early feasibility trades complete**

- Airlock
- Deep Space Transport



Deep Space Gateway
(DSG) Concept

Orion

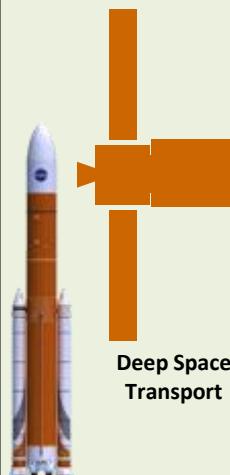
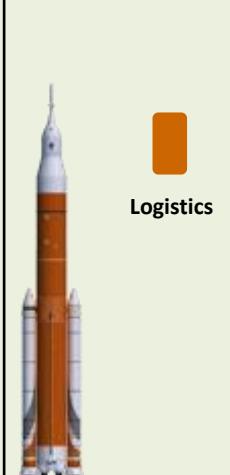
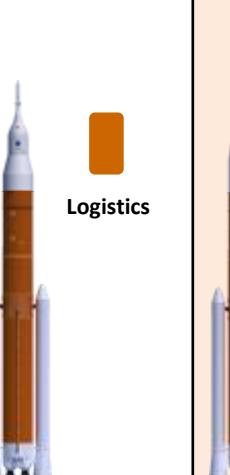
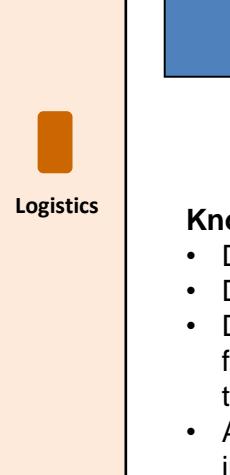
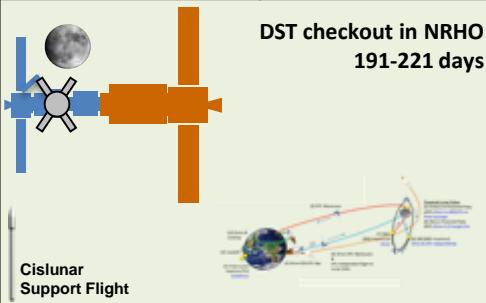
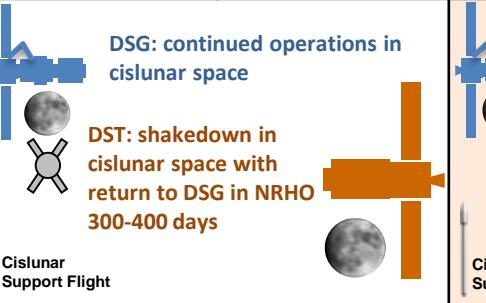
Deep Space
Transport (DST) Concept

PHASE 2

(PLANNING REFERENCE) Phase 2 and Phase 3

Looking ahead to the shakedown cruise and the first crewed missions to Mars



Transport Delivery		Transport Shakedown		Mars Transit	
EM-6	EM-7	EM-8	EM-9	EM-10	EM-11
2027		2028 / 2029		2030+	
SLS Block 1B Cargo P/L Capability: 41t TLI 	SLS Block 1B Crew: 4 CMP Capability: 10t 	SLS Block 1B Cargo P/L Capability: 41t TLI 	SLS Block 2 Crew: 4 CMP Capability: 13+t 	SLS Block 2 Cargo P/L Capability: 45t TLI 	SLS Block 2 Crew: 4 CMP Capability: 13+t 
Deep Space Transport	Logistics	DST Logistics & Refueling	Logistics	DST Logistics & Refueling	Logistics
Cislunar Support Flight	DST checkout in NRHO 191-221 days 	DSG: continued operations in cislunar space Cislunar Support Flight	DST: shakedown in cislunar space with return to DSG in NRHO 300-400 days 	DSG: continued operations in cislunar space Cislunar Support Flight	DST: Mars transit and return to DSG in NRHO 

Reusable Deep Space Transport
supports repeated crewed missions to the Mars vicinity

Known Parameters:

- DST launch on one SLS cargo flight
- DST shakedown cruise by 2029
- DST supported by a mix of logistics flights for both shakedown and transit
- Ability to support science objectives in cislunar space

Open Opportunities:

- Order of logistics flights and logistics providers
- Shakedown cruise vehicle configuration and destination/s
- Ability to support lunar surface missions

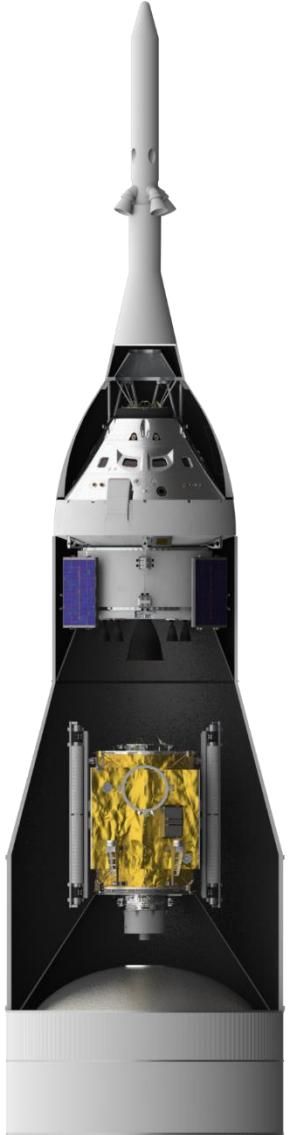


To inform decision making of DSG&T consistent with a delivery of the Power/Propulsion Element (PPE) for launch as a co-manifested payload on EM-2, HEOMD is leading the development and maturation of the following products through the fall/winter of 2017:

- High level **Exploration Objectives**. Initially baselined in September 2016; updated in June 2017. Establishes 60 objectives in phases 0, and 1, and 2 timeframes. Discussions ongoing with international partners for opportunities to align U.S. and partner deep space exploration objectives.
- **Interoperability Standards** that provide a comprehensive set of HEOMD-level requirements necessary to certify the DSG&T for deep space missions.
- **Exploration Requirements** and **Design Concept of Operations** that captures the HEO-level requirements and design conops for the systems we are building to implement Exploration Objectives. Evolving Exploration Systems Development requirements to include DSG&T requirements and reflect SLS and Orion support for DSG&T buildup.
- **Utilization Plan** that defines how and when we will use the systems, mission by mission, and what capabilities and resources are provided to support civilian, commercial, and international partner utilization of the DSG&T.

To ensure consistent coordination and communication across all exploration activities, these products are configuration controlled at the HEOMD level consistent with the **HEOMD configuration management process**.

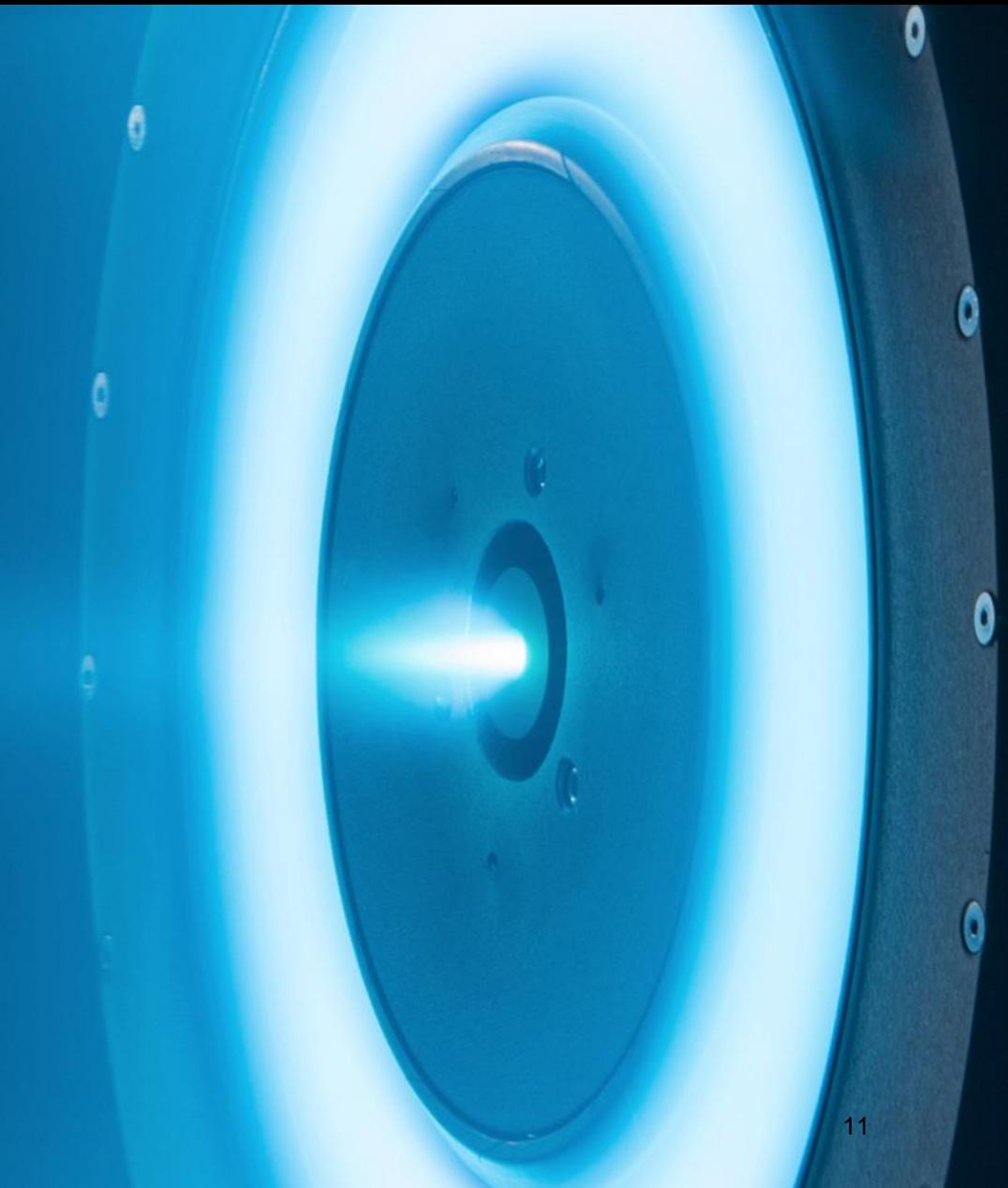
- Start deep space gateway when we fly crew to vicinity of the moon
- A power and propulsion element (PPE) would be the first element in a cis lunar gateway
- The PPE would provide key functionality for the DSG including
 - Power to DSG and externally accommodated elements
 - Transportation for the DSG between cis lunar orbits
 - Orbital maintenance as needed
 - Attitude control for the DSG in multiple configurations with and without visiting vehicles such as Orion
 - Communications with Earth, space to space communications, and radio frequency relay capability in support of extra-vehicular activity (EVA) communications.
- PPE will launch co-manifested with Orion crew vehicle on the Space Launch System for the EM-2 flight



Approach to PPE Development



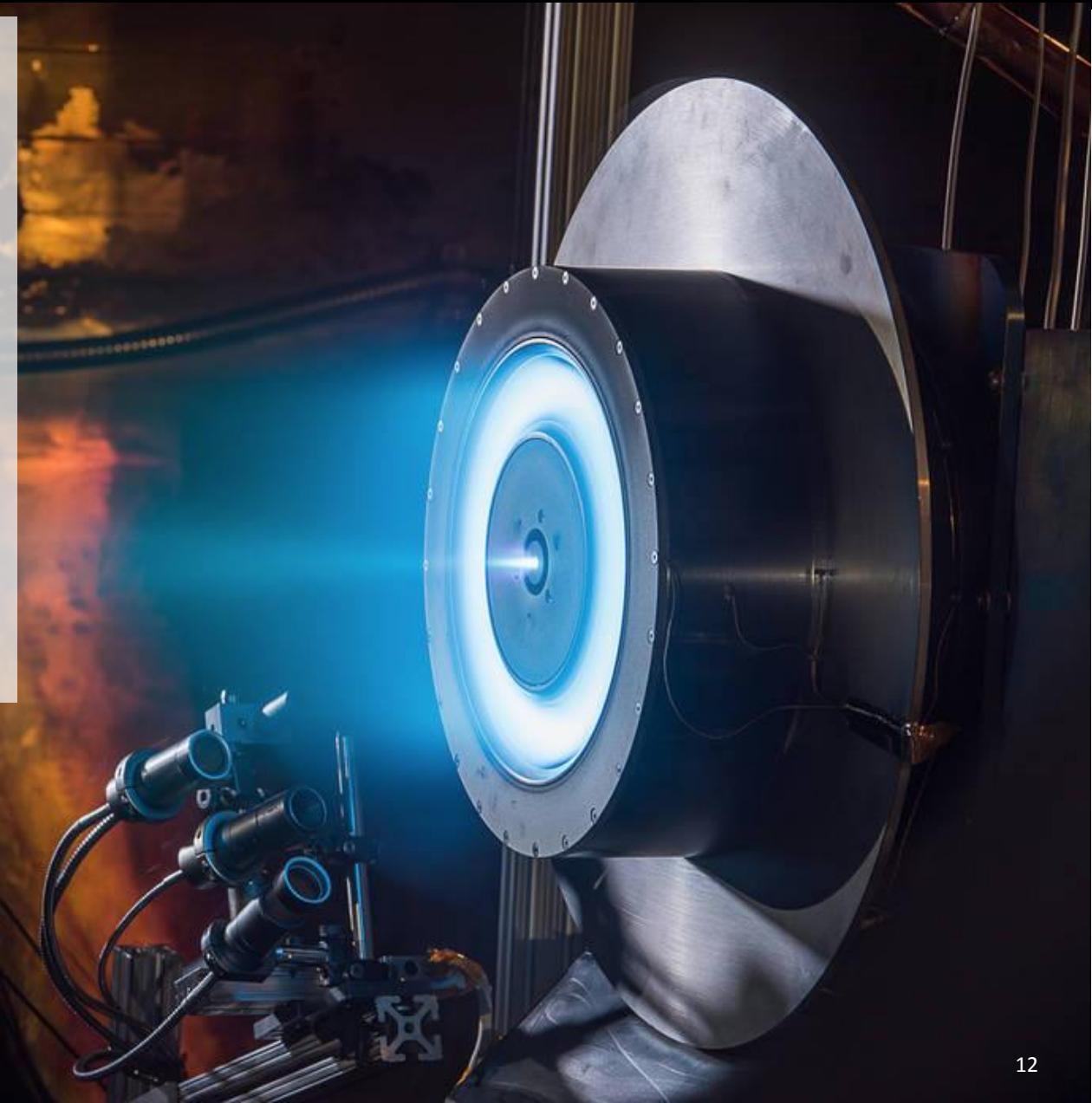
- **PPE will leverage advanced solar electric propulsion (SEP) technologies developed and matured during Asteroid Redirect Mission activities:**
 - Directly use commercially available U.S. flight hardware
 - Infuse advanced SEP technology developed by NASA's Space Technology Mission Directorate
 - Align with U.S. industry plans for future use of SEP
 - Accommodate international and/or commercial partner provided capabilities



Advanced Electric Propulsion Development



- Led by Glenn Research Center with JPL support
- Developed 13 kW Hall thruster
 - with magnetic shielding to extend lifetime
 - Isp: 2000-3000 sec
- Three test development units (TDU's) fabricated and tested (TDU-1 shown here)
- Integrated power testbed being assembled at GRC
- Vendor for engineering development and flight unit delivery under contract - Aerojet Rocketdyne
- Development to be completed by end of 2018
- Flight unit delivery by 2019



✓ Update stakeholders on Asteroid Redirect Mission transition and potential plan for solar electric propulsion	Jul2017
✓ Release of PPE Request for Information (RFI)	Jul 17, 2017
✓ Synopsis for Studies through NextSTEP BAA Appendix C	Jul 17, 2017
✓ RFI Responses Due	Jul 28, 2017
✓ STMD Electric Propulsion System Preliminary Design Review	Aug 1-3, 2017
✓ Draft PPE Industry Studies Solicitation Release	Aug 11, 2017
✓ PPE virtual industry day	Aug 17, 2017
✓ Discussions with industry	Aug 17-18, 2017
✓ Last day for written questions and comments on the draft release	Aug 22, 2017
✓ PPE Control Board: Establish Reference Requirements	Aug 24, 2017
✓ Deep Space Gateway Control Board: PPE Top Level Reference Requirements	Aug 28, 2017
✓ Final Release, NextSTEP BAA Appendix C: Power & Propulsion Studies	Aug 30, 2017
• Study Proposals Due	Sep 26, 2017

